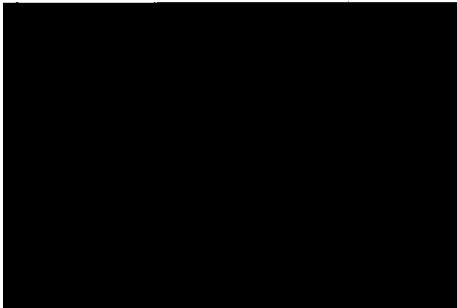




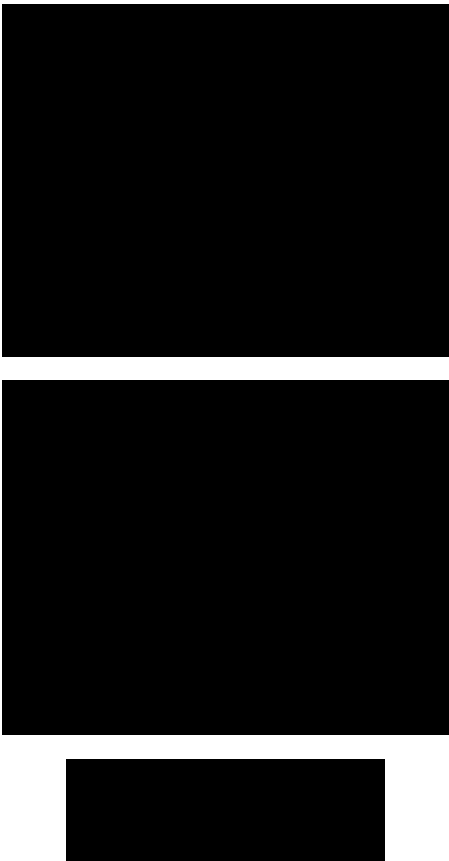
airfoil using PROFOIL in a few seconds and complete the com-



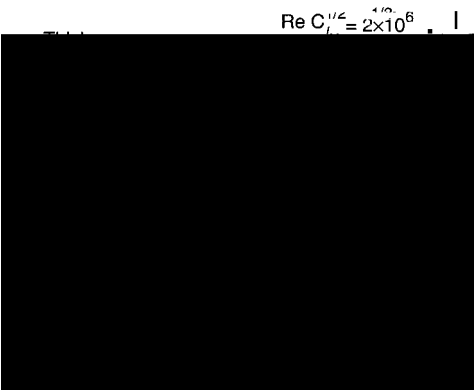
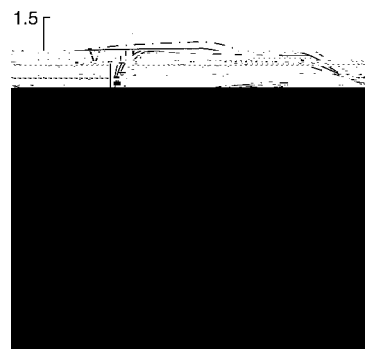
— A  
- - - B Re = 9×10<sup>6</sup>



upper surface to achieve a desired maximum thickness, while si-







at a  $C_l$  of 0.4. Examination of the velocity

